

Depth Control of Autonomous Underwater Vehicle Using Discrete Time Sliding Mode Controller

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Abstract This study presents a Discrete Time Sliding Mode Controller (DSMC) application on depth plane of Autonomous Underwater Vehicle (AUV). The main contribution on this work is an implementation of DSMC on NSP AUV II. Sliding Mode Control (SMC) is a robust type of controller and certainly suitable for controlling AUV in the presence of environmental disturbances and uncertainties. DSMC preserves the properties of standard SMC. Linearized dynamic model of NSP AUV II is used in the numerical simulations. Discrete Proportional Integral Derivative (PID) controllers are used for performance comparative analysis. The design of discrete PID and DSMC for NSP AUV II depth is described. Comparative study between the control laws is presented. The simulated results illustrate strong robustness, improve performance and satisfactory stability of DSMC as compared to discrete-time PID controller.

Keywords Autonomous Underwater Vehicle, Depth Control, Discrete Sliding Mode Controller, Discrete Proportional-Integral-Derivative

1. Introduction

Autonomous Underwater Vehicle (AUV) has shown popularity for three decades due to its versatility and excellent performance which is increasingly being used in many industries [1]. Their solid small size with self-operated propulsion systems, capability carrying sensors such as depth sensors, video cameras, side-scan sonar and other oceanographic measuring devices makes AUV be well suited in dangerous mission. Futuristic element in AUVs prompts advantage to wider area such as surveillance, environmental monitoring, underwater inspection of harbor and pipeline, geological and biological survey, mine counter measures and so forth. However, an extremely unexpected ocean behavior created challenges to

AUV navigation and motion performance in which this phenomenon demonstrates highly frequency oscillating movement that not only affects the sensor performance especially acoustical and optical sensors but also creates the dynamics system into highly nonlinear, time-varying and uncertainties in hydrodynamic parameters such as added mass, lift forces, gravity and buoyancy forces [2]. Additionally, most AUVs are operated on under actuated mode, hence tracking and stabilization control become demanding task, owing to over possession degree of freedom (DOF) beyond control [3]. Furthermore, this limitation is imposed in real life application due to inverting or pointing vertically that can cause equipment damage or dangerous control response [4]. As a result, AUVs motion control is restricted to only one noninteracting subsystem at a time [5]. Due to aforementioned challenges, many advanced control techniques have been implemented in existing literatures, mostly including robust control techniques in [6]–[8] intelligent control method in [9] and adaptive control approach in [10]–[12]. It's apparent that among robust controller types, SMC evidently considers a promising strategy [13] to overcome above obstacles.

The work reported in the literature addresses, the majority of the SMC application on AUV is in continuous time point of view but their effectiveness in real situations cannot be guaranteed [13]. This is because, all of these systems are operated on the discrete time domain in practice with utilization on digital computers or microprocessors, hence continuous time control cannot be implemented [13]. As a result, discrete time sliding mode control (DSMC) has produced significant interest over recent years [13]–[15] in solving the problems caused by the discretization of continuous time controllers. Started in 1997, Lee [16] adopted self-tuning discrete sliding mode control on AUV ARMA based on equivalent discrete variable structure control method and continue research on quasi sliding mode control in presence of uncertainties and long sampling interval in [17] on AUV VORAM. Followed

by Shaowei Zhang [18] proposed discrete-time quasi sliding mode controller for the multiple-input multiple-output on AUV REMUS. Lastly, Baoju [19] implemented adaptive sliding mode control in discrete time system and applied time varying sliding surface obtained by parameter estimation method.

Research in discrete-time controller designed has started by Dote and Hoft [20] that introduced reaching condition. Later Sarpturk, *et al* [21] revised Dote and Hoft reaching condition and founded equivalent form based on Lyapunov method. Then, Furuta [22] proposed another reaching condition on the basis of Lyapunov function. Finally, quasi sliding mode band was created by Gao [23] based on non-switching reaching condition and Bartoszewics [24] based on non-switching condition. Although Gao's reaching law method has been introduced since two decades ago, it's still been used in many significant studies such as [25]–[27].

The main aim on this research is to implement reaching law proposed in [26] on depth motion control. Discrete PID and DSMC using reaching law by Gao's in [28] are tested on AUV NSP II via simulation. Discrete PID controller is

used for performance comparative analysis. The paper is organized as follow: Section 2 introduced dynamic model of AUV NSP II in the Body-Fixed Reference frame (BFF). Section 3 presents discrete sliding mode control structure designed. Results from numerical simulation are illustrated in section 4 and section 5 providing discussion on advantages of the control methods.

2. Mathematical Model of AUV NPS II

2.1. Nonlinear General Equation of Motion

According to Fossen [29], general motion of AUV can be described by using BFF and earth-fixed frame (EFF) demonstrated in Figure 1. The parameters are defined according to [30] as shown in Table 1. The element of motion in BFF is given in the following vectors.

$$v = [v_1 \ v_2]^T \tag{1}$$

$$v_1 = [u \ v \ w]^T \text{ Linear velocities} \tag{2}$$

$$v_2 = [p \ q \ r]^T \text{ Angular velocities} \tag{3}$$

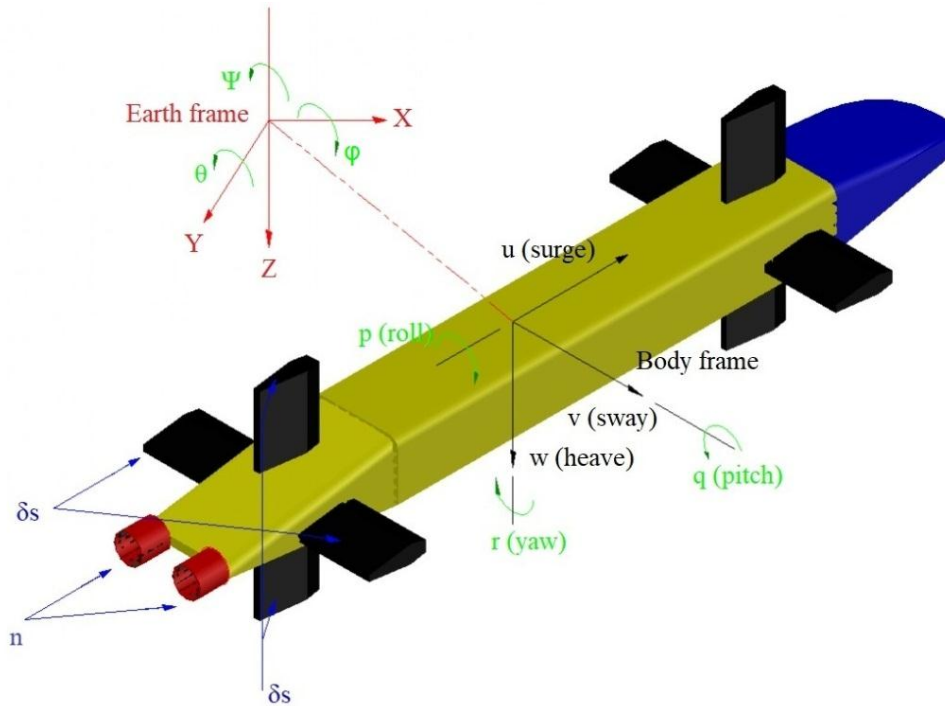


Figure 1. The schematic of NSP II BFF and EFF [29]

Table 1. The Notations for marine vessel [30]

Degree of Freedom	Forces & Moments	Linear and Angular Velocities	Position and Euler Angles
Surge	X	u	x
Sway	Y	v	y
Heave	Z	w	z
Roll	K	p	ϕ
Pitch	M	q	θ
Yaw	N	r	ψ

The position and attitude vectors of the BFF with referring to EFF express as:

$$n = [n_1 \ n_2]^T \quad (4)$$

$$n_1 = [x \ y \ z]^T \text{ Position of Origin} \quad (5)$$

$$n_2 = [\phi \ \theta \ \psi]^T \text{ Angles orientation of roll } (\phi), \quad (6)$$

pitch (θ) and yaw (ψ)

The nonlinear dynamic equation of motion in the BFF is defined as follows

$$M\dot{v} + C(v)v + D(v)v + G(\eta) = \tau \quad (7)$$

Where, $M \in \mathfrak{R}^{6 \times 6}$ is in inertia matrix, $C(v) \in \mathfrak{R}^{6 \times 6}$ is Coriolis and centripetal matrix, $D(v) \in \mathfrak{R}^{6 \times 6}$ is damping matrix, $G(\eta) \in \mathfrak{R}^{6 \times 1}$ is vector of buoyancy/gravitational forces/moments matrix and $\tau \in \mathfrak{R}^{6 \times 1}$ is vector of control inputs relating to forces and moments acting on vehicle.

The control input vector τ has three components as:

$$\tau = [\delta_r, \delta_e, n] \quad (8)$$

Where δ_e is elevator deflection, δ_r is rudder deflection and n is propeller revolutions.

2.2. Linearized equation of motion

The kinematic equation of motion in depth plane is given by:

$$\dot{z} = -u(\sin\theta) + w(\cos\theta) \quad (9)$$

$$\dot{\theta} = q$$

u and w representing surge and heave velocities of the vehicle. In steady state motion of a vehicle, $\theta_0 = q_0 = \phi_0 = 0$. Hence kinematic equation is rewritten as:

$$\dot{z} = -\theta u_0 + w \quad (10)$$

$$\dot{\theta} = q$$

The depth plane dynamics equation of motion is obtained by setting all state vector related to steering plane to zero ($v=p=r=\phi=\psi=0$). Hence, equation of motion in depth plane express as:

$$m(\dot{w} - u_0 q) = Z \quad (11)$$

$$I_y \dot{q} = M$$

The external forces X , Z and moment M described by hydrodynamic added mass, linear damping and the effects

of the elevator deflection, depth plane model is simplified as:

$$Z = Z_w \dot{w} + Z_q \dot{q} + Z_w w + Z_q q + Z_\delta \delta_s$$

$$M = M_w \dot{w} + M_q \dot{q} + M_w w + M_q q - mg(z_G - z_B) \sin\theta + M_\delta \delta_s \quad (12)$$

From (10), (11) and (12), linearized model in vertical plane equation of motion can be written in the following form.

$$\begin{bmatrix} m - Z_{\dot{w}} & -Z_{\dot{q}} & 0 & 0 \\ -M_{\dot{w}} & I_y - M_{\dot{q}} & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} \dot{w} \\ \dot{\theta} \\ \theta \\ \dot{z} \end{bmatrix} + \begin{bmatrix} -Z_w & mu_0 - Z_q & 0 & 0 \\ -M_w & -M_q & (Z_G - Z_B)W & 0 \\ 0 & -1 & 0 & 0 \\ -1 & 0 & u_0 & 0 \end{bmatrix} \begin{bmatrix} w \\ q \\ \theta \\ z \end{bmatrix} = \begin{bmatrix} Z_\delta \\ M_\delta \\ \theta \\ z \end{bmatrix} \delta_s \quad (13)$$

As in [31], the heave velocity during diving is small (less than 0.05m/s), thus term containing w and \dot{w} can be neglected. Since w and \dot{w} are neglected, the mathematical model of vertical plane as in (13) can be expressed in state space form as:

$$\begin{bmatrix} \dot{q} \\ \dot{\theta} \\ \dot{z} \end{bmatrix} = \begin{bmatrix} \frac{M_q}{I_y - M_{\dot{q}}} & \frac{-W(Z_G - Z_B)}{I_y - M_{\dot{q}}} & 0 \\ 1 & 0 & 0 \\ 0 & -u_0 & 0 \end{bmatrix} \begin{bmatrix} q \\ \theta \\ z \end{bmatrix} + \begin{bmatrix} \frac{M_\delta}{I_y - M_{\dot{q}}} \\ 0 \\ 0 \end{bmatrix} \delta_s \quad (14)$$

Where M_q is pitch moment due to q , $M_{\dot{q}}$ is pitch moment due to time rate of change of q , I_y is vehicle inertia around the pitch axes, W is AUV weight, Z_G is center of gravity, Z_B is center of buoyancy, u_0 is designed velocity and M_δ is fin lift coefficient. Then, (14) can be expressed as:

$$\dot{x}(t) = Ax(t) + Bu(t) \quad y(t) = Cx(t) \quad (15)$$

3. Depth Control of AUV NPS II in DSMC

DSMC is designed in this section to control desired depth of NSP II AUV. Considering discrete-time linear system described by

$$(k+1) = (k) + \Gamma u(k) \quad (16)$$

$$(k+1) = (k) \quad (17)$$

Where $x(k) \in \mathfrak{R}^n$ is the state vector, $u(k) \in \mathfrak{R}^m$ is the control input, $\Phi \in \mathfrak{R}^{n \times n}$ and $\Gamma \in \mathfrak{R}^{n \times n}$ are the system matrices.

The sliding surface in discrete time is given by

$$S(k) = C_s e(k) = C_s (x_r(k) - x(k)) \quad (18)$$

Where $e(k)$ is the tracking error, $x_r(k)$ is reference input and $C_s \in \mathfrak{R}^m$ is the sliding matrix selected such that $C_s \Gamma$ is nonsingular.

The system is designed to steer the state trajectory to the origin when travelling along switching surface. Without loss of generality, the following linear sliding surface is defined as:

$$S(k) = 0 \tag{19}$$

Considering (18), the first-time derivative of the sliding surface is given by:

$$S(k + 1) - S(k) = C_s[x_r(k + 1) - x(k + 1)] - C_s[x_r(k) - x(k)] \tag{20}$$

Substituting (16) into (20), the first derivative of sliding surface is rewritten as:

$$S(k + 1) - S(k) = C_s[x_r(k + 1) - \Phi x(k) - \Gamma u(k)] - C_s[x_r(k) - x(k)] \tag{21}$$

A discrete-time extension of reaching law approach was proposed by Gao [32].The reaching law in this case is given by

$$S(k + 1) - S(k) = -qTS(k) - \varepsilon Tsgn(S(k)) \tag{22}$$

Where $\varepsilon > 0, q > 0$ and $1 - qT > 0$ and T is the sampling time.

Hence, the control law is expressed as:

$$u(k) = -(C_s\Gamma)^{-1}[-C_s x_r(k + 1) + C_s\Phi x(k) + C_s x_r(k) - C_s x(k) - qTs(k) - \varepsilon Tsgn(s(k))] \tag{23}$$

Following steps are to obtain sliding gain matrix C_s .

Firstly, by substitute (22) into (23), it yields

$$X(k + 1) = (\Phi - \Gamma K)X(k) \tag{24}$$

$$\text{Where } K = (C_s\Gamma)^{-1}C_s\Phi \tag{25}$$

Hence, the sliding gain matrix C_s becomes the solution of the following equation.

$$C_s(\Phi - \Gamma K) = 0 \tag{26}$$

$$C_s\Gamma = I \tag{27}$$

Where I is an identity matrix and (27) is to ensure that $C_s\Gamma$ is full rank. Using (26), (27) can be replaced by $C_s\Phi = K$ and thus above equations can be rewritten as:

$$C_s[\Phi - \Gamma] = [K I] \tag{28}$$

Finally, the sliding matrix C_s is given by

$$C_s = [K I][\Phi \Gamma]^+ \tag{29}$$

Where $+$ is representation of matrix pseudo-inverse. The feedback matrix K can be obtained by adopting (27) into LQR controller [1].

4. Simulation on Depth Control

To demonstrate effectiveness of discrete sliding mode controller, simulation has been made on AUV NSP II specifically on depth control system. A depth control is simulated with the full linear six degree of freedom (DOF) and simplified equation of motion reduced to heave z and pitch θ . The AUV NSP II is 5.3m in length. Its weight in air is 5443.4 kg and has natural buoyancy in water. The flooded mass in water is about 53400N. The maximum speed in calm water is 3m/s [33]. Added mass inertia, $M_{\dot{q}}$ is -1.7×10^{-2} and moment of inertia in y -axes, I_y is 13587 nms^2 .

To illustrate an effectiveness of DSMC, discrete PID controller is used as a comparative analysis. Step response simulations are performed in heave and pitch DOF at $T=0.2$. PID controller is widely used due to its reliability and simplicity but it's difficult to tune the parameters in discrete PID controllers to achieve optimal performances. PID gain setting as in Table 2.

Table2. PID Gain value

Gain	Value
K_p	-0.1629
K_d	-4.0117e-05
K_i	3.09681

Using (29), the sliding matrix C_s is given by

$$C_s = [1.6136 \quad -1.9098 \quad -0.225] \tag{30}$$

The reaching law parameters are selected as follow

$$q = 0.4, \varepsilon = 0.01 \tag{31}$$

Dynamic model of AUV NSP II for diving control was carried out by setting up the desired depth is 8m. Fig.2 shows result of diving control responded under ideal conditions. The settling time taken is 180s and 250s for DSMC and discrete PID respectively. DSMC proved shortest settling time due to its sensitivity to parameter variations. Fig.3 indicates elevator deflection commanded by the controller to AUV as an input and AUV pitch angle response as output shown in Fig.4. The result illustrated elevator deflection response by DSMC is smoother and sensible. In contrast to discrete PID controller, elevator response shows lumpy motion to pitch angle of AUV. From this simulation, it's clearly demonstrated that the proposed method improved controller performances, strong robustness and satisfactory stability.

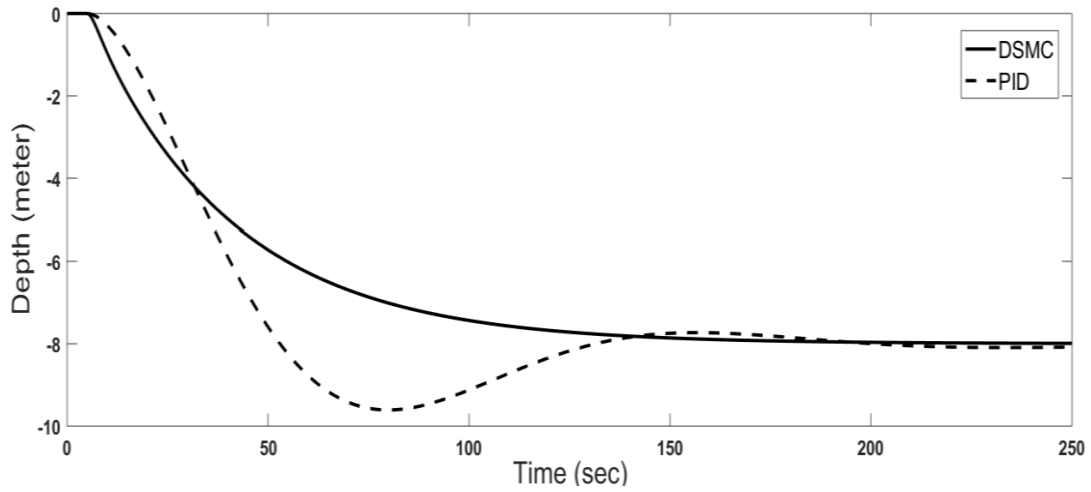


Figure 2. Vehicle depth response

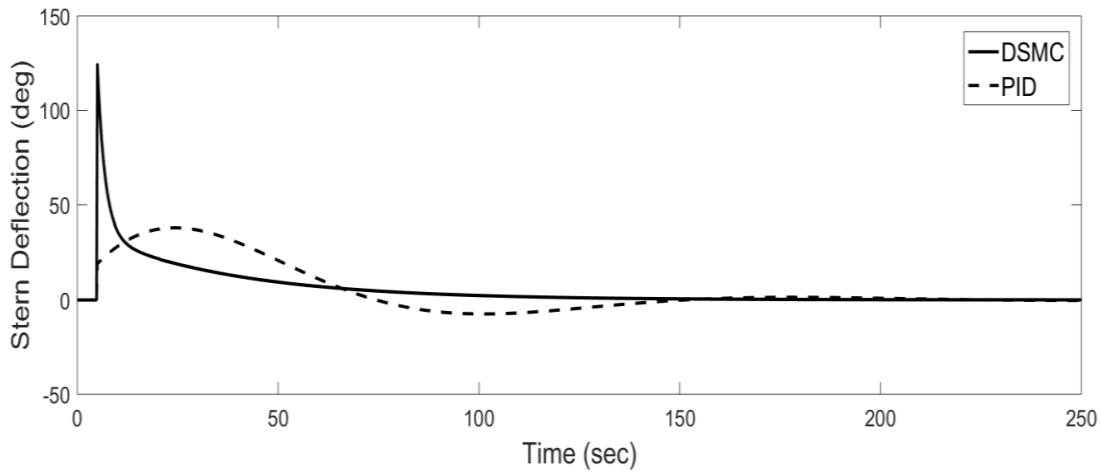


Figure 3. Vehicle elevator deflection response

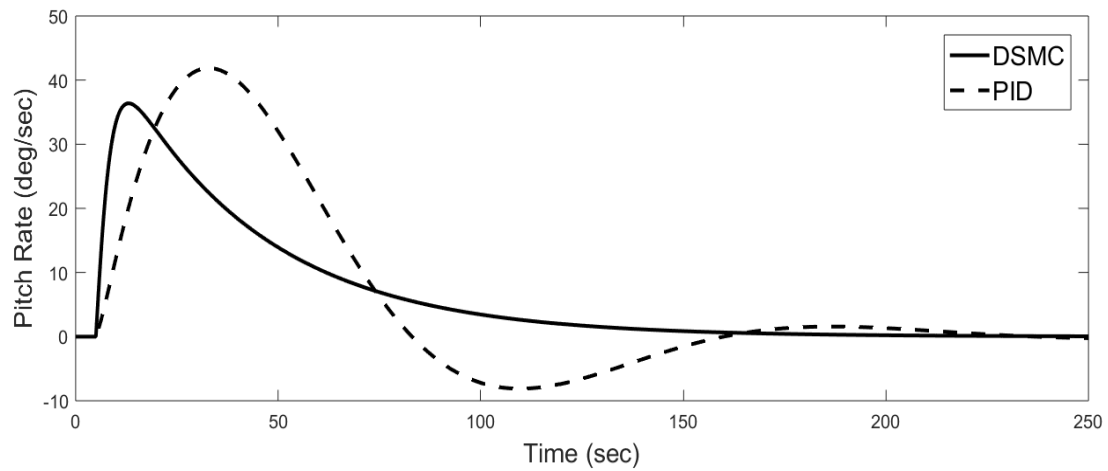


Figure 4. Vehicle pitch rate response

5. Conclusions

In this study, the algorithm of discrete SMC was proposed. The reason of choosing is due to robustness against external disturbance and uncertain dynamics. Finally, the simulation and experiment result show excellent performance in depth control of NSP AUV II. Two control methods for depth regulation consisting of discrete SMC and discrete PID and each performance was compared via simulation. The result indicates discrete-time SMC has higher control precision, faster convergence and stronger robustness than discrete-time PID. In the future work, disturbance will be added into algorithm and both controller performances will be tested via simulation and experiment.

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